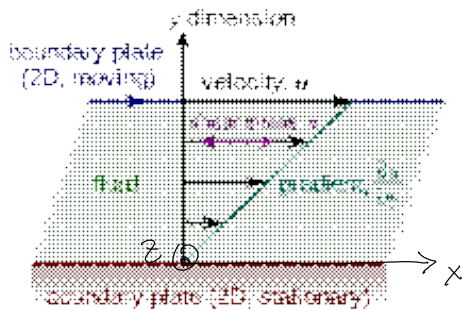


# L04 Damped motion

Monday, September 24, 2018 04:01

- Linear Drag: Viscosity - <https://en.wikipedia.org/wiki/Viscosity>  
See also: PPT Purcell, "Life at Low Reynolds Number"  
and [https://en.wikipedia.org/wiki/Pitch\\_drop\\_experiment](https://en.wikipedia.org/wiki/Pitch_drop_experiment)



"coefficient of friction in a fluid"

$$\tau_{xy} \equiv F_x / A_y = F_x / dx dz$$

$$= \eta \frac{\Delta u}{\Delta y}$$

shear stress      viscosity

Taylor ch2 #2.2.

STP:  $\eta = 1.7 \times 10^{-5} \text{ N}\cdot\text{s}/\text{m}^2$   
 $\beta = 1.6 \times 10^{-4} \text{ N}\cdot\text{s}/\text{m}^2$

See Taylor, footnote 8, p72:  $F = \eta A dv/dy$        $f_{\text{lin}} = \underbrace{3\pi\eta D}_{b} v$  (sphere)

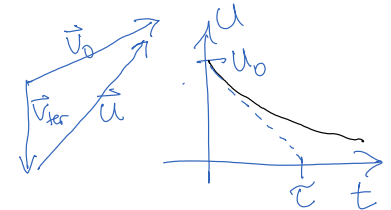
\* 3d solution to linear damping in gravity:

$$\vec{F} = m\vec{\dot{v}} = \underbrace{m\vec{g}}_{b\vec{v}_{\text{ter}}} - b\vec{v} \quad \text{let } \vec{u} = \vec{v} - \vec{v}_{\text{ter}} \quad \vec{\dot{u}} = \vec{\dot{v}} \quad b\vec{v}_{\text{ter}} = m\vec{g}$$

$$m\vec{\dot{u}} = -b\vec{u} \quad \text{let } \tau = \frac{m}{b} \quad \vec{u} = \hat{u} u$$

$$\int_{u_0}^u \frac{du}{u} = \int_0^t -\frac{dt}{\tau}$$

$$\ln u/u_0 = -t/\tau \quad u = u_0 e^{-t/\tau}$$



$$(\vec{v} - \vec{v}_{\text{ter}}) = (\vec{v}_0 - \vec{v}_{\text{ter}}) e^{-t/\tau} \quad \vec{v}_{\text{ter}} = \frac{m\vec{g}}{b} = \tau\vec{g} \quad \tau = m/b$$

$$\vec{r}(t) = \vec{r}_0 + \int_{t=0}^t \vec{v}(t) dt = \boxed{\vec{r}_0 + \vec{v}_{\text{ter}} t - (\vec{v}_0 - \vec{v}_{\text{ter}}) \tau (e^{-t/\tau} - 1)}$$

- trajectory: let  $\vec{r}_0 = (0,0)$   $\vec{v}_0 = (v_{x0}, v_{y0})$ , note  $\vec{v}_{\text{ter}} = (0, v_{\text{ter}})$

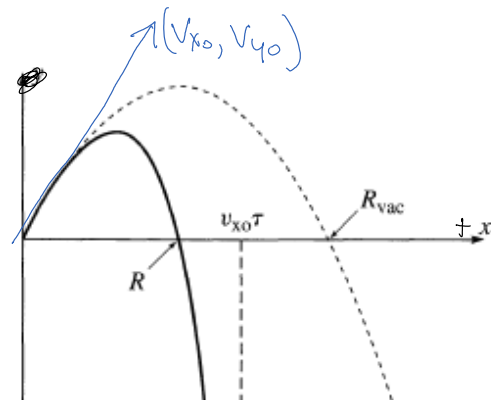
$$v_x = v_{x0} e^{-t/\tau} \quad x = \underbrace{x_0}_{=0} + \int_0^t v_x dt = v_{x0} (-\tau) e^{-t/\tau} \Big|_0^t = v_{x0} \tau (1 - e^{-t/\tau})$$

$$1 - e^{-t/\tau} = \frac{x}{v_{x0} \tau} \quad t = -\tau \ln(1 - \frac{x}{v_{x0} \tau})$$

$$v_y = v_{\text{ter}} + (v_{y0} - v_{\text{ter}}) e^{-t/\tau} \quad [+ \text{ going down}]$$

$$y = \int_0^t v_y dt = \underbrace{y_0}_{=0} + v_{\text{ter}} t + (v_{y0} - v_{\text{ter}}) \tau (1 - e^{-t/\tau})$$

$$= v_{\text{ter}} (-\tau \ln(1 - \frac{x}{v_{x0} \tau})) + (v_{y0} - v_{\text{ter}}) \tau (1 - \frac{x}{v_{x0} \tau})$$



$$= v_{ter} \left( -\tau \ln \left( 1 - \frac{x}{v_{ter} \tau} \right) \right) + (v_{y0} - v_{ter}) \tau \left( \frac{x}{v_{ter} \tau} \right)$$

$$= \frac{v_{y0} - v_{ter}}{v_{x0}} x - v_{ter} \tau \ln \left( 1 - \frac{x}{v_{ter} \tau} \right) \quad +y \downarrow$$

$\frac{x \approx 0}{\rightarrow} \frac{v_{y0} x}{v_{x0}} \quad , \quad \frac{x \approx v_{ter} \tau}{\rightarrow} -\infty$

Fig 2.7 p55, Taylor

- Quadratic drag: the force required to accelerate the air in front to  $v$

$$dm = \rho dV = \rho A v dt$$

$$F = (ma = dm \frac{dv}{dt}) \cdot \kappa$$

$$\kappa = 1/4 \text{ "streamlined"}$$

$$\rho = 1.29 \text{ kg/m}^3 \text{ STP}$$

$$\gamma = 0.25 \text{ N}\cdot\text{s}^2/\text{m}^4$$

$$f_{quad} = \kappa \rho A v^2 = \underbrace{\kappa \rho \frac{\pi}{4} D^2}_c v^2 \quad (Taylor \text{ ch2 \#2.4})$$



\* 1d solution for quadratic drag

$$F = \underbrace{c v_{ter}^2}_{\text{drag}} - \underbrace{c v^2}_{\text{drag}} = m \underbrace{\dot{v}}_{\text{dv/dt}} \quad \text{let} \quad mg = c v_{ter}^2$$

$$\dot{v} = g \left( 1 - \frac{v^2}{v_{ter}^2} \right)$$

$$\frac{dv}{1 - v^2/v_{ter}^2} = g dt$$

$$v_{ter} \frac{\text{sech}^2 \alpha d\alpha}{\text{sech}^2 \alpha} = g dt$$

$$v_{ter} \alpha = gt$$

$$v = v_{ter} \tanh \left( \frac{gt}{v_{ter}} \right)$$

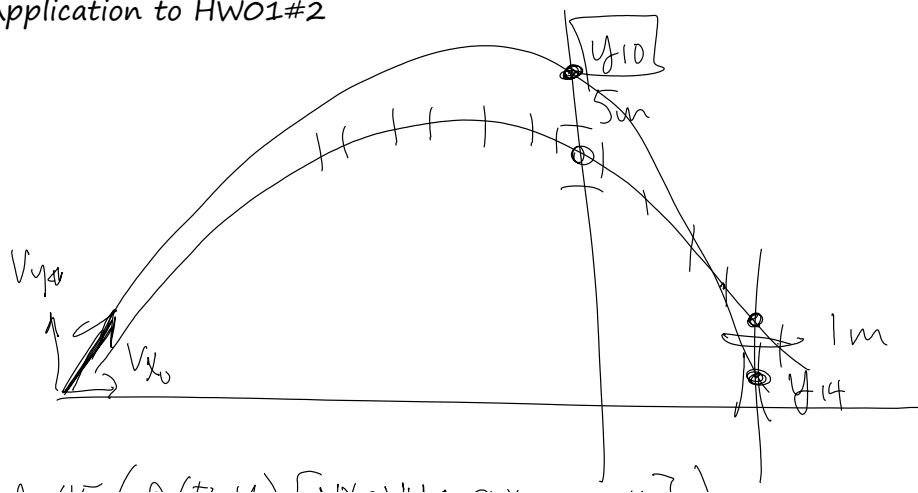
$$\text{let } v = v_{ter} \tanh \alpha$$

$$\cosh^2 \alpha - \sinh^2 \alpha = 1$$

$$1 - \tanh^2 \alpha = \text{sech}^2 \alpha$$

$$dv = v_{ter} \cdot \text{sech}^2 \alpha \cdot d\alpha$$

- Application to HWO1#2



---


$$\text{ode45}(\text{@}(t,u)[v_x; v_y; g_x, g_y], u(3), u(4))$$

$$\vec{F} = m\vec{a} = m\vec{g} - cv^2\hat{v}$$

$$u = \begin{bmatrix} x \\ y \\ v_x \\ v_y \end{bmatrix} \quad \vec{a} = \vec{g} - cv\vec{v} \quad \sqrt{\vec{v} \cdot \vec{v}} = v^2 = v$$

$$a_x = g_x - c\sqrt{v \cdot v} v_x$$

$$= g_x - c \text{sqrt}(\text{dot}(u(3:4), u(3:4))) * u(3)$$